

⁶Institute of Energy and Climate Research, IEK-8: Troposphere, Forschungszentrum Jülich GmbH, Jülich, Germany

^anow at: School of Environmental Science and Engineering, Gwangju Institute of Science and Technology, Gwangju, Korea

^bnow at: Department of Chemistry, University of York, York, UK

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Correspondence to: R. A. Washenfelder (rebecca.washenfelder@noaa.gov)

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Abstract

We describe a two-channel broadband cavity enhanced absorption spectrometer (BBCEAS) for aircraft measurements of glyoxal (CHOCHO), methylglyoxal (CH₃COCHO), nitrous acid (HONO), nitrogen dioxide (NO₂), and water (H₂O). The instrument spans 361–389 and 438–468 nm, using two light emitting diodes (LEDs) and a grating spectrometer with a charge-coupled device (CCD) detector. Robust performance is achieved using a custom optical mounting system, high power LEDs with electronic on/off modulation, state-of-the-art cavity mirrors, and materials that minimize analyte surface losses. We have successfully deployed this instrument during two aircraft and two ground-based field campaigns to date. The demonstrated precision (2σ) for retrievals of CHOCHO, HONO and NO₂ are 34, 350 and 80 pptv in 5 s. The accuracy is 5.8, 9.0 and 5.0 %, limited mainly by the available absorption cross sections.

1 Introduction

Broadband cavity enhanced absorption spectroscopy (BBCEAS) belongs to a class of techniques that use high-finesse optical cavities to achieve sensitive measurements of optical extinction (Fiedler et al., 2003). BBCEAS is distinct from other techniques in this class, such as cavity ring-down spectroscopy (CRDS), because it employs a broadband light source and a multichannel detector. Broadband measurements enable the simultaneous detection of multiple absorbing species across a wide spectral region, and the quantification of species with significant spectral overlaps in their features. The basic components of BBCEAS include a light source, collimating optics, high-finesse cavity with two high reflectivity mirrors, and a multi-channel detector to monitor wavelength-dependent light changes inside of the cavity.

Laboratory studies have used BBCEAS to measure a number of trace gases with structured absorption in the visible and ultraviolet spectral region, including NO₂, NO₃, N₂O₅, HONO, H₂O, H₂O₂, O₃, O₄, I₂, IO, OIO, SO₂, CHOCHO, and CH₃COCHO

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(Ball et al., 2004; Venables et al., 2006; Gherman et al., 2008; Vaughan et al., 2008; Washenfelter et al., 2008; Langridge et al., 2009; Watt et al., 2009; Thalman and Volkamer, 2010; Axson et al., 2011; Chen and Venables, 2011; Kahan et al., 2012), as well as wavelength-dependent aerosol extinction (Thompson and Spangler, 2006; Washenfelter et al., 2008, 2013; Thalman and Volkamer, 2010; Varma et al., 2013). Ground- and ship-based BBCEAS field instruments have been used to measure CHOCHO and NO₂ (Washenfelter et al., 2011a; Coburn et al., 2014), HONO and NO₂ (Washenfelter et al., 2011a; Young et al., 2012), NO₃, and N₂O₅ (Benton et al., 2010). Kennedy et al. (2011) have recently described an aircraft BBCEAS instrument to measure NO₂, NO₃ and N₂O₅. Based on this history, glyoxal (CHOCHO) and nitrous acid (HONO) are promising targets for aircraft BBCEAS measurements. Aircraft measurements of trace gases are important to satellite validation; characterization of chemical lifetimes, processing, and transport; quantification of emissions and deposition; and the study of chemical mechanisms.

Glyoxal is the simplest alpha-dicarbonyl, and is one of the most prevalent dicarbonyls in the ambient atmosphere. Global models indicate that its major source is oxidation of biogenic compounds, led by isoprene (47 %) with smaller contributions from monoterpenes (4 %) and methylbutenol (0.8 %) (Fu et al., 2008). Other important glyoxal sources include the oxidation of anthropogenic species, such as ethyne and aromatics (30 %), and direct production from biomass burning (18 %) (Fu et al., 2008). However, these global estimates are uncertain, in part because of the limited number of ambient measurements of glyoxal. For example, enhanced glyoxal concentrations over the equatorial ocean have recently been reported from satellite and ship-based instruments, indicating a source of CHOCHO from the oxidation of marine emissions (Sinreich et al., 2010; Mahajan et al., 2014). The major loss processes for glyoxal include photolysis, reaction with OH and NO₃, and deposition to aerosol and ground surfaces. Laboratory studies have reached different conclusions about the magnitude and reversibility of glyoxal's contribution to secondary organic aerosol mass (Jang et al., 2002; Hastings et al., 2005; Kroll et al., 2005; Liggio et al., 2005; Nakao

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et al., 2012), while field studies have also drawn inconsistent conclusions about the importance of glyoxal to aerosol mass budgets (Volkamer et al., 2007; Washenfelder et al., 2011a). Previously, remote sensing measurements of ambient glyoxal concentrations have been reported using long-path differential optical absorption spectroscopy (DOAS) (Volkamer et al., 2005a; MacDonald et al., 2012) and multi-axis DOAS (MAX-DOAS) (Baidar et al., 2013). The few successful techniques for in situ measurements include high-pressure liquid chromatography or gas chromatography (Munger et al., 1995; Grosjean et al., 2002), laser-induced phosphorescence (Huisman et al., 2008), BBCEAS (Washenfelder et al., 2011a), and the related cavity-enhanced DOAS (CE-DOAS) technique (Coburn et al., 2014). The uncertainty in both the sources and sinks of glyoxal underscore the need for rapid, accurate, in situ measurements.

Similar to glyoxal, there is a need for accurate, rapid, in situ measurements of HONO, which plays an important role in atmospheric radical budgets through photolytic production of OH radicals and NO (Platt et al., 1980; Alicke et al., 2002). Scientific questions remain about its sources, sinks, and vertical profile (Zhou et al., 1999; Su et al., 2008; Young et al., 2012; VandenBoer et al., 2013). Existing detection methods can be categorized as wet chemistry, mass spectrometry, and optical spectroscopy. The wet chemical detection methods are sensitive, but generally rely on conversion of HONO to nitrite ion (NO_2^-) (Appel et al., 1990; Dibb et al., 2002; Kleffmann et al., 2006) and may be susceptible to chemical interferences and sampling artifacts (Stutz et al., 2010). Recently, chemical ionization mass spectrometry (CIMS) with acetate ion (Roberts et al., 2010) or iodide ion (Veres et al., 2015) chemistry has been successfully used for sensitive HONO detection. Spectroscopic methods for remote sensing include long-path DOAS and MAX-DOAS (Platt et al., 1980; Hendrick et al., 2014). In situ, spectroscopic detection methods include cavity ring down spectroscopy (Wang and Zhang, 2000), tunable diode laser spectroscopy (Li et al., 2008; Lee et al., 2011) and Fourier transform spectroscopy (Barney et al., 2000; Yokelson et al., 2007), although limited attempts have been made for field deployment. BBCEAS has been used successfully

for ground-based measurements of HONO (Washenfelter et al., 2011a; Young et al., 2012).

Here, we present a new aircraft BBCEAS instrument, the Airborne Cavity Enhanced Spectrometer (ACES), and describe its use to measure CHOCHO, HONO, NO₂, methylglyoxal (CH₃COCHO), and water (H₂O). This instrument follows the development of laboratory and ground-based BBCEAS field instruments by our group (Washenfelter et al., 2008, 2011a; Young et al., 2012), with significant improvements in engineering and data acquisition that allow rapid, precise aircraft sampling. This is the first instrument for in situ measurements of CHOCHO from an aircraft. The HONO measurement is lower in precision than that of CHOCHO due to lower cavity mirror reflectivity and narrower LED spectral output, but it is sufficient for aircraft measurements under high signal conditions, such as in biomass burning plumes. The ACES instrument was successfully deployed during the SouthEast NEXus (SENEX) 2013 and the Shale Oil and Natural Gas Nexus (SONGNEX) 2015 aircraft studies, where it operated on the NOAA WP-3D aircraft at altitudes from 0–7 km and flight durations of 6–7 h. In addition, ACES has been deployed to measure ambient CHOCHO and HONO during two ground-based field campaigns: the Campaigns of Air Quality Research in Beijing and Surrounding Regions in the North China Plain (CARE Beijing-NCP) 2014 and the Uintah Basin Winter Ozone Study (UBWOS) 2014. Section 2 describes the instrument in detail. We present data from the SENEX 2013 and CARE Beijing-NCP 2014 studies in Sects. 3 and 4.

2 Experimental description of the ACES instrument

The ACES optical system, flow system, electronics, cell materials, and operation are described in detail in this section. The instrument builds on our previous ground-based field instrument (Washenfelter et al., 2011a), with numerous improvements to achieve robust performance in a demanding flight environment subject to vibration and rapid changes in ambient temperature and pressure.

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The ACES instrument has two channels, as shown in Fig. 1a, with wavelength coverage from 361–389 and 438–468 nm (hereafter Ch 368 and Ch 455). The wavelength range is determined by the LED spectral radiance, the center wavelength and bandwidth of the cavity mirrors, as well as the wavelength-dependent absorption features of the target gases (Fig. 2). HONO and NO₂ are detected in Ch 368, while CHOCHO, CH₃COCHO, NO₂, and H₂O are detected in Ch 455. In the future, the instrument could be expanded to detect other trace gases with absorption features in these spectral regions, such as IO, or other trace gases in different spectral regions by replacing the LEDs and cavity mirrors.

2.1 Optical system

The optical system is shown schematically in Fig. 1a and technical details for the components are given in Table 1. Two LEDs (NCSU033B, Nichia Corp., Tokyo, Japan; LZ1-00DB05, LedEngin Inc., San Jose, CA, USA) are separately temperature-controlled using thermoelectric coolers. Their output light is collimated using off-axis parabolic mirrors (50328AL, 2.0 cm effective focal length, Newport Corp., Irvine, CA, USA) prior to entering each cell. The off-axis parabolic mirrors optimize both photon throughput and space efficiency. The light is coupled into a high-finesse cavity formed by high-reflectivity mirrors (1 m radius of curvature, Advanced Thin Films, Boulder, CO, USA), which are separated by 48 cm. The difference in sensitivity between this design and a confocal mirror separation is negligible (Wild et al., 2014).

The light exiting the cavity is imaged by an off-axis parabolic mirror (50331AL, 15.2 cm effective focal length, Newport Corp., Irvine, CA, USA) through a band pass filter onto a 0.5 cm F/2 lens that couples the light into a 1 m UV-VIS fiber optic bundle. The bundle contains two groups of seven 200 μm-diameter fibers (BFB-455-7, Princeton Instruments, Trenton, NJ, USA) that are linearly aligned on the slit axis of a 150 mm focal length f/4.0 grating spectrometer (SP2150, Princeton Instruments, Trenton, NJ, USA). The CCD detector has 2048 × 512 pixels and a –70 °C cooling capacity (PIXIS2KBUV, Princeton Instruments, Trenton, NJ, USA). The spectrometer

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contains a 1200 groove mm^{-1} (500 nm blaze) grating centered at 418 nm, with spectral coverage of 119 nm. Two regions of interest (2048×128 pixels each) are defined based on the illumination of the CCD. In addition, a dark region of the CCD (2048×50 pixels) between the two regions of interest is recorded to monitor photon diffusion among adjacent pixels and light scattering inside of the spectrometer.

The CCD must be darkened while the accumulated charge is moved across the semiconductor surface to a readout amplifier and analog-to-digital converter. This would typically be achieved using a mechanical shutter that requires ~ 8 ms to open and close, and can fail after rapid, continuous operation. To improve our instrument duty cycle and reliability, we electronically modulate the optical output of the LEDs. We further improve the duty cycle and signal-to-noise ratio with a low readout amplification (gain: $16 \text{ e}^- \text{ count}^{-1}$, digitization speed: 2 MHz) and a CCD temperature of -50°C .

For automated wavelength and lineshape calibrations, we use a Hg lamp (HG-1, Ocean Optics, Dunedin, FL, USA) with a custom adapter located between the band pass filter and fiber optic bundle to introduce Hg light without obscuring the main beam path. A fused silica window (Edmund Optics, Barrington, NJ, USA) mounted at 45° inside the adaptor reflects Hg light to illuminate the fiber optic during wavelength calibrations. The spectrometer wavelength calibration and lineshape are calculated from four narrow Hg lines at 365.02, 404.66, 407.78 and 435.83 nm (air wavelengths; Sansonetti et al., 1996), assuming a Gaussian lineshape.

2.2 Cage system and hardware

As shown in Fig. 1a, the optical components are mounted in a cage system, measuring $63 \times 29 \times 13$ cm. The cage system consists of carbon fiber rods with 1.25 cm outside diameter (OD), and mounts for each optical component attached to the rods (Wild et al., 2014). Custom-designed aluminum plates aligned and locked parallel to each other on the carbon rods provide robust alignment without the use of spring-loaded, commercial mirror mounts. This system provides stable optical alignment that does not require re-

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alignment and is insensitive to vibration, pressure, or temperature changes. The optical cavity, spectrometer, and CCD are mounted in a temperature-controlled box, which is maintained at 30 or 35 °C using thermoelectric coolers.

2.3 Flow system

The flow system for ACES includes the inlet, aerosol filter, two cells, pressure sensors, temperature sensors, mass flow controllers, and pump. It is shown schematically in Fig. 1b and described in further detail below.

The inlet consists of two parts: a coaxial inlet designed for overflow by calibration gases and a filter to remove aerosols. All parts are made from Teflon perfluoroalkoxy polymer resin (PFA), which provided the best CHOCHO transmission efficiency in laboratory tests (see Sect. 2.4). The coaxial inlet consists of 0.95 cm OD tubing (0.35 m length, 0.79 cm ID), which contains 0.64 cm OD tubing (0.15 m length). The 0.64 cm OD tubing runs through the PFA tee to allow additions of calibration gases (Wagner et al., 2011). The calibration gas flow is 0.3–0.5 sL min⁻¹ (standard liter per minute) greater than the total sample flow rate of 5.0 sL min⁻¹. During aircraft sampling, the total length of the coaxial inlet is 1.5 m and it is contained in a winglet that extends outside the boundary layer of the aircraft.

A PTFE membrane filter (25 µm thickness, 4.6 cm diameter, 1 µm pore size; Tisch Environmental, Cleves, OH, USA.) is installed downstream of the coaxial inlet to remove ambient aerosol because aerosol extinction limits the effective path length and adds a variable background extinction. The filter is changed prior to each 8 h aircraft flight and once or twice per day for ground based measurements. Ground-based data were examined for discontinuities in CHOCHO or HONO before and after a filter change, and no discrepancies were observed within the measurement precision. Prior experience with NO₂ (Osthoff et al., 2006; Wagner et al., 2011) and CHOCHO (Washenfelder et al., 2011a) suggest that aerosol filters transmit these compounds quantitatively.

After the filter, the flow is evenly divided to two sample cells constructed of Teflon PFA (2.5 cm OD, 1.5 cm ID, 40 cm length). Unlike in previous designs (Wagner et al.,

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2011; Washenfelder et al., 2013), the ACES instrument does not include mirror purges of dry gas for mirror cleanliness. The mirror purges were found to be unnecessary, and their absence eliminates the small uncertainties from flow dilution and the relative sample length occupied by sample gas and purge gas over the length of the cavity.

5 The cells are connected to separate mass flow controllers (MC-50SLPM-D-DB15B, Alicat Scientific, Inc., Tucson, AZ, USA) and a scroll pump (IDP-3, Agilent Technologies, Inc., Santa Clara, CA, USA) to maintain a constant flow rate of 2.5 sL min^{-1} , resulting in a residence time less than 1.5 s in each cell. This flow rate and residence time were consistent with the data acquisition rate and minimized He and ZA consumption
10 during reference measurements. The cell pressures (PPT, Honeywell International Inc., Plymouth, MN, USA) and temperatures (KMQSS-020U-6, Omega Engineering Inc., Stamford, CT, USA) are measured immediately downstream of the flow.

The cavity loss (or effective optical path length) was measured by sequentially over-
15 flowing the inlet with He and zero air (ZA) from compressed gas cylinders or by acquiring zero air measurements at different pressures. Further details are given in Sect. 3.1.

2.4 Flow system materials

Flow system materials can potentially cause production or loss of target analytes, affecting the accuracy of in situ measurements. Previous tests have demonstrated that NO_2 has negligible losses on Teflon and metal surfaces (Osthoff et al., 2006; Fuchs
20 et al., 2009). Prior measurements have shown that inlet length has negligible impact on glyoxal losses for PTFE tubing (Huisman et al., 2008).

We measured glyoxal losses for ten materials to determine the best cell and tubing choices. For the tests, we configured both BBCEAS channels to measure CHOCHO, with the cells connected in series and a length of tubing between them. Losses be-
25 tween the two cells were attributed to the tubing material. Constant CHOCHO concentrations were generated from a temperature-controlled Teflon bubbler (Saville, Eden Prairie, MN, USA) filled with CHOCHO solution (40 weight percent in water; Sigma Aldrich, St. Louis, MO, USA). Each test section was $1.0 \pm 0.1 \text{ cm}$ ID and $92 \pm 2 \text{ cm}$

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long. Table 2 lists the ten tubing materials and their measured losses, in units of fractional loss per unit tubing surface area per residence time ($\% \text{cm}^{-2} \text{s}^{-1}$). Teflon PFA showed the smallest loss ($0.0001 \pm 0.005 \% \text{cm}^{-2} \text{s}^{-1}$), while glass tubing had greatest loss ($0.211 \pm 0.006 \% \text{cm}^{-2} \text{s}^{-1}$). Finishing methods also affected the loss rate, with lower losses by polished metal surfaces compared to unpolished metal surfaces. We calculate the total loss of CHOCHO from the surface area and residence time of individual components of the ACES sample system to be less than 0.1 %, even for inlet tubing of 10 m in length and 0.79 cm ID with 5 sL min^{-1} flow. The relative humidity during these experiments was low ($< 20 \%$) and constant. At elevated relative humidity, the losses may be greater, although we expect this to be a small effect (Washenfelter et al., 2008) and did not observe a dependence of the CHOCHO/HCHO ratio on relative humidity during SENEX 2013 that would indicate inlet effects. Actual inlet lengths were 1.5 m for aircraft sampling during SENEX and SONGNEX, 3–12 m for ground-based sampling during CARE Beijing and 6.5 m for ground-based sampling during UBWOS 2014.

2.5 Operation of the ACES instrument

During aircraft operation, the ACES instrument is typically turned on 1 h prior to flight, allowing the LEDs, optics box, and CCD to stabilize in temperature. The CCD dark background is characterized by acquiring 50 spectra under dark conditions with the same integration time as the subsequent measurements. This is necessary because the CCD produces non-zero signal under optically dark conditions. At the beginning and end of each flight, we measure dark background spectra, Hg spectra for wavelength calibration and instrument lineshape, and standard additions of NO_2 and CHOCHO. The inlet filter is changed before and after each flight.

During flight, ZA is introduced for 30 s every 5 min, while He is introduced for 30 s every 15 min for calibrating mirror reflectivity. During ground-based operation, the gas additions are 30–60 s in duration, depending on the inlet length. Inlet filter change, dark background, and Hg wavelength and instrument lineshape measurements are performed once or twice per day.

3 Data analysis

3.1 Determination of cavity loss and trace gas extinction

BBCEAS instruments measure steady-state light intensity transmitted through an optical cavity. The light attenuation inside the cavity includes (1) absorption, scattering, and transmission losses by the cavity mirrors, (2) Rayleigh scattering by gas within the cavity, (3) absorption by trace gases, and (4) Mie scattering by particles. A general expression describing the sum of the extinction is

$$\sum_i^n \alpha_i(\lambda) = R_L \left(\frac{(1 - R(\lambda))}{d} + \alpha(\lambda)_{\text{Rayleigh}} \right) \left(\frac{I_0(\lambda) - I(\lambda)}{I(\lambda)} \right) \quad (1)$$

where $\alpha_i(\lambda)$ is extinction by the i th component, λ is the wavelength of light, R_L is the ratio of the total cell length to the sample length, d is the cavity length, $R(\lambda)$ is the mirror reflectivity, $I_0(\lambda)$ is the reference spectrum (without absorbing trace gases or aerosols and at the same temperature and pressure as the sample spectrum), and $I(\lambda)$ is the measured spectrum of ambient air, including trace gas absorptions, as a function of wavelength (Washenfelter et al., 2008).

The ACES instrument has no mirror purges, so R_L is unity. We define $I_0(\lambda)$ as the reference zero air spectrum so that the term $\alpha(\lambda)_{\text{Rayleigh}}$ explicitly includes its Rayleigh scattering. For aircraft measurements, we must also explicitly account for the difference between the Rayleigh scattering of the reference zero air spectrum, $I_{\text{ZA}}(\lambda)$, and sample spectrum, $I_{\text{sample}}(\lambda)$, due to pressure differences during sampling. Incorporating these changes gives

$$\sum_i^n \alpha_{i,\text{sample}}(\lambda) = \left(\frac{(1 - R(\lambda))}{d} + \alpha_{\text{Ray,ZA}}(\lambda) \right) \left(\frac{I_{\text{ZA}}(\lambda) - I_{\text{sample}}(\lambda)}{I_{\text{sample}}(\lambda)} \right) + \Delta\alpha_{\text{Ray}}(\lambda) \quad (2)$$

where $\Delta\alpha_{\text{Ray}}(\lambda) = \alpha_{\text{Ray,ZA}}(\lambda) - \alpha_{\text{Ray,sample}}(\lambda)$. The $\alpha_{i,\text{sample}}(\lambda)$ summation on the left hand side of Eq. (2) includes only absorption by gas-phase species, because aerosol parti-

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cles are eliminated by the Teflon inlet filter. This term can be rewritten as

$$\sum_i^n \alpha_{i,\text{sample}}(\lambda) = \sum_i^n \sigma_i(\lambda) N_i \quad (3)$$

where σ_i and N_i are the absorption cross section and number density for the i th trace absorber.

Accurate measurements of the trace gas absorption by BBCEAS require calibration of mirror reflectivity, $R(\lambda)$, or the cavity loss, which represents the inverse of the effective path length and is defined as $\alpha_{\text{cavity}}(\lambda) = (1 - R(\lambda))/d$. This quantity can be determined by introducing a species with well-known extinction into the cavity. For example, previous studies have used known Rayleigh scattering cross sections of He and N₂ or ZA (Washenfelter et al., 2008; Thalman and Volkamer, 2010) or N₂ and N₂/NO₂ mixtures (Langridge et al., 2006; Venables et al., 2006). We have tested and compared two methods to determine $\alpha_{\text{cavity}}(\lambda)$. First, we have compared Rayleigh scattering extinction in ZA relative to that in He (referred to as He/ZA in the following text). Second, we have used ZA spectra acquired at different known pressures (referred to as ZA/ZA in the following text). The second method is possible because the ACES cage system maintains its optical alignment during pressure changes, and has the advantage of eliminating the need for He gas cylinders. To our knowledge, this method has not been reported previously.

Empirical expressions for Rayleigh scattering cross section were determined using fits to Bodhaine et al. (1999) for N₂ ($\sigma_{\text{Rayleigh},\text{N}_2} = 1.2577 \times 10^{-15} \times \lambda^{-4.1814}$); Shardanand and Rao (1977) and Snee and Ubachs (2005) for O₂ ($\sigma_{\text{Rayleigh},\text{O}_2} = 1.0455 \times 10^{-15} \times \lambda^{-4.1814}$); and Shardanand and Rao (1977) for He ($\sigma_{\text{Rayleigh},\text{He}} = 1.336 \times 10^{-17} \times \lambda^{-4.1287}$). Rayleigh scattering cross sections recently reported by Thalman et al. (2014) were not used in the calculations here, but will be evaluated for future retrievals.

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3.2 Determination of trace gas concentrations

Both published and measured reference spectra have been used to fit the number density of each absorber according to Eq. (3). For the literature absorption cross sections, high resolution laboratory reference were used for NO₂ (Vandaele et al., 2002), CHO-CHO (Volkamer et al., 2005b), CH₃COCHO (Staffelbach et al., 1995), H₂O (Harder and Brault, 1997), O₄ (Greenblatt et al., 1990), and HONO (Stutz et al., 2000).

Literature reference spectra were convolved with a Gaussian lineshape of full-width at half-maximum (FWHM) determined from Hg calibration lines, which was 0.82 nm for Ch 368 and 1.02 nm for Ch 455. The convolved literature reference spectra are shown in Fig. 2 for the relevant wavelength regions. Measured reference spectra were determined from NO₂ and CHOCHO additions as described in Sect. 4.4, and these were used for the Ch 455 spectral retrievals.

The number density of trace gases was determined using least-squares, DOAS-style fit retrievals (Platt and Stutz, 2008) with DOASIS fitting software (Kraus, 2006). The fitted absorbers were HONO, NO₂, and O₄ for Ch 368, and CHOCHO, CH₃COCHO, NO₂, H₂O and O₄ for Ch 455. In addition, a 3rd- or 4th-order polynomial was included in each fit to account for drift in the light intensity and cavity throughput of the measurement. Stretch and shift of the reference spectra wavelength were included as fit parameters to minimize discrepancies with the wavelength calibration of the grating spectrometer, which may be due to physical shifts in the spectrometer optics. For Ch 368, the measured NO₂ reference spectra was shifted and stretched to minimize the fit residual, with equivalent values prescribed for the HONO and O₄ reference spectra. For Ch 455, the measured NO₂ and CHOCHO reference spectra were shifted and stretched to minimize the fit residual. The literature H₂O, CH₃COCHO, and O₄ reference spectra were shifted and stretched separately.

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4 Results and discussion

4.1 Measured cavity loss

The pressure response of the ACES instrument was tested by measuring N_2 extinction at sample pressures between 300–600 hPa in the laboratory, as shown in Fig. 3a. The measured optical extinction from Eq. (1) at 455 nm is plotted against the number density, N , determined from temperature and pressure. $I_0(\lambda)$ is defined to be the spectra acquired at 400 mb. Similarly, the slope at each wavelength element can be used to determine the Rayleigh scattering cross section of N_2 at that wavelength, and the results for 438–468 nm are shown in Fig. 3b, with comparison to Bodhaine et al. (1999). The direct comparison of $\sigma_{\text{Rayleigh}, N_2}$ from ACES and Bodhaine et al. (1999) show good agreement (slope = 0.993 ± 0.007 , $r^2 = 0.993$) indicating the pressure-insensitive performance of ACES.

Figure 4 compares the cavity loss determined from the He/ZA and ZA/ZA methods during the SENEX 2013 research flight on 5 July 2013. He and ZA were measured 13 times during level flight legs, at pressures between 390 to 812 mb. The calculated cavity losses, $\alpha_{\text{cavity}}(\lambda) = (1 - R(\lambda))/d$, using He/ZA at 368 and 455 nm, where HONO and CHOCHO absorption cross section are maximized, are $(3.534 \pm 0.016) \times 10^{-6}$ and $(5.551 \pm 0.017) \times 10^{-7} \text{ cm}^{-1}$ respectively (all values given as mean $\pm 1\sigma$). The α_{cavity} calculation from the ZA/ZA method, with dry zero air measurements at 390 and 812 mb, falls within the variation of that from the He/ZA method at these two wavelengths. This comparison is additional evidence of the pressure-insensitive performance of ACES, which enables robust performance under varying external pressure. It also demonstrates that mirror reflectivity calibrations could be carried out using pressure differences alone, without the need for He gas.

Example plots showing measured intensity, cavity loss due to Rayleigh scattering, effective path length, and calculated detection limit for one pair of He/ZA measurements made on 5 July 2013 are shown in Fig. 5. For Ch 368, the signal intensity of the

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LED has a maximum near 368 nm, while the mirror reflectivity is optimal near 376 nm. Ch 455 has maximum LED intensity much closer to the best mirror reflectivity near 460 nm, and close to the maximum glyoxal absorption feature at 455 nm. Intensity in Fig. 5 is plotted as CCD detector counts for a 0.5 s exposure, and the actual photon count rate is $16\times$ greater due to the detector gain. Rayleigh scattering loss is comparable to cavity loss for Ch 368, but is much smaller than cavity loss for Ch 455. The effective path length, neglecting Rayleigh scattering, is 3.0 km at 368 nm and 17.8 km at 455 nm. The lower panels show the minimum detectable extinction as a function of wavelength on each channel as a function of wavelength, calculated from the photon shot noise limit. See caption for further details.

4.2 Spectral fitting

Figures 6 and 7 show spectral fitting examples from the 3 July 2013 flight data during SENEX. Figure 6 shows simultaneous retrievals of NO_2 , HONO and O_4 with a 3rd-order polynomial at 361–389 nm, along with the fit result and fit residual. Figure 7 shows simultaneous retrievals of NO_2 , CHOCHO , CH_3COCHO , and H_2O with a 4th-order polynomial at 438–468 nm, along with the total fit result and fit residual. No systematic patterns in the fit residual are observed, indicating that there are no systematic errors with wavelength or missing gas-phase absorbers. The increased noise in the residuals at the edge of each fit window is due to decreased LED power (reduced photon counts) and decreased mirror reflectivity. These examples are from intercepts of a biomass burning plume with large absorption by CHOCHO and HONO, as well as measurable CH_3COCHO .

4.3 Precision and accuracy

The instrumental precision and stability can be determined from optical extinction during continuous zero air measurements. Figure 8 shows an Allan deviation plot (Allan, 1966) for optical extinction at two individual pixels, calculated from 1 h of data acquired

at constant pressure. The Allan deviation (2σ) follows the square root of the averaging time up to few hundred seconds. The Allan deviation in Ch 455 is roughly four times smaller than Ch 368 up to 1 min averaging time due to the longer effective light path length of Ch 455. The extinction values shown in Fig. 8 represent an upper limit for the precision of trace gas retrievals because they are calculated for single pixels, rather than the range of pixels used in spectral fitting.

Figure 9 shows spectral retrievals of NO_2 , CHOCHO , and HONO concentrations fit to zero air spectra acquired during ground measurements in the CARE Beijing-NCP 2014 study. The acquisition time for these data is $0.478 \text{ s point}^{-1}$ with an 84 % data cycle (CCD exposure time: 0.4 s, compensation time: 0.006 s, and readout time: 0.072 s).

The instrumental precision (2σ) for each trace absorber is estimated from two different methods: (1) standard deviation from dry zero air retrievals and (2) fit errors derived from the DOAS retrieval (Platt and Stutz, 2008). For both methods, the data sets for the Allan deviation analysis and regular zero air measurements during the field mission (30 s injection at 5 min intervals) were used, as described above. The estimated precision (2σ) for each retrieved trace gas is summarized in Table 3. Precisions are adequate for ambient measurement of CHOCHO , HONO , and NO_2 , although the current precision for HONO is insufficient to quantify the small ambient mixing ratios reported during daytime or in low NO_x ($\text{NO} + \text{NO}_2$) environments (e.g. Ren et al., 2010).

The accuracy of the retrieved trace gas concentrations depends on the uncertainty of the literature reference cross sections, Rayleigh scattering cross sections, pressure, temperature, and inlet losses. The uncertainties in absorption cross sections are: 4 % for NO_2 (Vandaele et al., 2002); 5 % for CHOCHO (Volkamer et al., 2005b); 15 % for CH_3COCHO (Staffelbach et al., 1995); 8.7 % for HONO (Stutz et al., 2000); 4 % for H_2O (Harder and Brault, 1997); and 10 % for O_4 (Greenblatt et al., 1990). The estimated uncertainty for the Rayleigh scattering cross sections of zero air is 2 % and the uncertainty for He makes a negligible contribution (Washenfelder et al., 2008). The uncertainties in pressure and temperature measurements are 0.5 and 0.7 %. CHOCHO loss along the inlet line is negligible, as described in Sect. 2.4 (< 0.1 % for any sam-

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pling conditions used for aircraft- and ground-based sampling). HONO inlet artifacts are more difficult to characterize, since HONO may undergo loss due to adsorption or production due to heterogeneous reaction of NO₂ and H₂O (Finlayson-Pitts et al., 2003). The latter is of particular concern in assessing the accuracy low-level HONO mixing ratios during daytime (Li et al., 2014). For this work, we have minimized HONO sampling artifacts by using a short inlet with short residence time, but have not characterized the inlet behaviour under different atmospheric conditions. The propagated errors (summed in quadrature) are $\pm 4.6/\pm 5.0\%$ for NO₂ and $\pm 5.5/\pm 5.8\%$ for CHO-CHO using literature/measured absorption cross sections and ± 15.2 , ± 9.0 , ± 4.6 , and $\pm 10.2\%$ for CH₃COCHO, HONO (neglecting potential inlet artifacts), H₂O, and O₄, respectively.

4.4 Standard additions of trace gases

4.4.1 NO₂

Standard NO₂ concentrations were generated using a custom calibration system (Washenfelter et al., 2011b), which generates and measures O₃ using a commercial ozone monitor (49i, Thermo Fisher Scientific Inc., Waltham, MA, USA) and subsequently titrates O₃ with excess NO (2040 ppm \pm 2%, NO in N₂, Scott-Marrin Inc., Riverside, CA, USA) to produce a known concentration of NO₂. During SENEX 2013, we used this system daily to generate standard NO₂ concentrations for validation, with typical NO₂ mixing ratios between 3–100 ppbv (parts per billion).

Figure 10a shows one set of standard additions from the NO₂ calibration system. The agreement between retrieved NO₂ from ACES and the NO₂ mixing ratio from the calibration system is shown in Fig. 10d. The relationship is linear over the entire range ($r^2 = 1.000$), with slopes (intercepts) of 1.031 ± 0.003 (0.16 ± 0.14 ppbv) and 1.015 ± 0.005 (0.23 ± 0.17 ppbv) for Ch 368 and Ch 455, respectively. The non-zero intercept is due to small concentrations of NO₂ which are present in the NO titration gas of the calibration system (Washenfelter et al., 2011b). The deviations of the measurement

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from the calibration source (3.1 and 1.5 %) are well within the absolute measurement accuracy of ± 5.0 % for NO_2 given in Sect. 4.2.

The NO_2 standard additions demonstrate the accuracy of the ACES instrument response. In addition, we performed standard additions of HONO and CHOCHO. For these additions, we compared the ACES measurement to the calculated flow dilution because the HONO and CHOCHO concentrations were not independently determined.

4.4.2 CHOCHO

Figure 10b shows five additions from the CHOCHO source, in the range 2–8 ppbv. Each addition was allowed to stabilize for 1 min, and the subsequent 2 min of data are shown in bold. Figure 10e shows the average and standard deviation for these additions. The correlation coefficient ($r^2 = 0.999$) indicates the linearity of the CHOCHO measurements from ACES and the CHOCHO calibration system. The retrieved fit errors from DOASIS were 30 pptv (parts per trillion; Table 3).

4.4.3 HONO

Constant HONO concentrations were generated using a calibration source based on the design in Roberts et al. (2010). Briefly, humidified air mixed with HCl from a permeation tube passes through a sodium nitrite bed (NaNO_2 mixed with glass beads) to generate HONO via acid displacement. The entire system (i.e. permeation tube, water vessel and NaNO_2 bed) is temperature-controlled for stable HONO generation. The output from the HONO generator may contain small amounts of NO and NO_2 .

Figure 10c shows three additions from the HONO source with different flow dilutions, in the range of 4–7 ppbv. Each addition was allowed to stabilize for 1 min, and the subsequent 2 min of data are shown in bold. The observed delay in measurement response after a change in concentration is due to the addition source and not the inlet, which responded rapidly (< 5 s) during aircraft sampling. Figure 10f shows the

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average and standard deviation of these additions, with high linearity ($r^2 = 0.999$). The DOASIS fit errors for the 5 s spectra were 314 pptv (Table 3).

4.5 Aircraft measurements during SENEX 2013

The ACES instrument was deployed on the NOAA WP-3D research aircraft during the SENEX project from 1 June to 15 July 2013, with flights based out of the Smyrna/Rutherford County Airport in Smyrna, Tennessee. In addition to the ACES instrument, many instruments to characterize gas- and aerosol-phase species were deployed (see partial list in de Gouw et al., 2015), including a CRDS instrument to measure NO, NO₂, NO₃, N₂O₅, and O₃ (Wagner et al., 2011). One goal of the field study was to understand the interactions between natural and anthropogenic emissions in the southeastern US and to evaluate their impact on air quality and climate.

Figure 11a shows CHOCHO measurements acquired on 5 July 2013 during a flight that sampled an isoprene hot spot (Ozark Mountains, MO), an urban area (St. Louis, MO), and an ethanol refinery (Decatur, IL). CHOCHO mixing ratios greater than 140 pptv were observed over St. Louis, with NO₂ mixing ratios greater than 14 ppbv. Outside of the urban area, measured CHOCHO concentrations were 53–79 pptv (interquartile range). The time series of CHOCHO is shown in Fig. 11c.

Time series data for NO₂ measured by ACES and CRDS are shown in Fig. 11b, with the CRDS data offset by 1 ppbv for clarity. The CRDS instrument reported NO₂ concentrations at 1 s time resolution with precision and accuracy of 200 pptv and 5 %, respectively. A scatter plot comparing the ACES Ch 455 and CRDS NO₂ data is shown in Fig. 11d, with the data averaged to 10 s. The instruments agree well, with slope of 0.983 ± 0.013 with $r^2 = 0.937$, which is consistent with the uncertainty of the measurements.

CH₃COCHO has been retrieved only for pyrogenic plumes during SENEX 2013, because its absorption cross section is small and contains fewer structured features

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than CHOCHO (see Fig. 2). As shown in Fig. 7, biomass burning plumes can represent large sources of CH_3COCHO , with measured mixing ratios up to 5.7 ppbv.

4.6 Ground-based measurements during CARE Beijing-NCP 2014

The ACES instrument was subsequently deployed during the CARE Beijing 2014 study from 2 June–6 July 2014. Figure 12a shows a map of the North China Plain field site, which was located 198 km southwest of Beijing in Wangdu, Hebei Province. The overall goals of the field study focused on HO_x radical chemistry, new particle formation, and the impacts of air pollution. The scientific goals for the ACES instrument included examining the importance of anthropogenic precursors in CHOCHO formation and characterizing the importance of CHOCHO and HONO as radical sources.

Figure 12 shows time series for NO_2 , HONO, CHOCHO, and H_2O acquired during six days in June 2014. Peak NO_2 and HONO concentrations were 65 and 3.5 ppbv at night, with peak concentrations of 240 pptv CHOCHO during the day. NO_2 and HONO mixing ratios were low during daytime and higher during nighttime, consistent with accumulation of NO_x emissions in a shallow nocturnal boundary layer and heterogeneous conversion of NO_2 to HONO on the ground surface. In contrast, CHOCHO concentrations had a morning maximum, which is not typical of its diurnal pattern in other locations (Volkamer et al., 2005a; Sinreich et al., 2007; Huisman et al., 2011; Washenfelder et al., 2011a). Previous studies have shown that CHOCHO concentrations track the OH oxidation of precursors, with a maximum in the afternoon, similar to other photochemically produced species. The unusual CHOCHO diurnal profile at this site suggests an interaction between emission, chemistry, and transport at the NCP site, such as a mixing of residual layer air masses with higher CHOCHO concentrations into the nighttime boundary layer during morning.

A number of other instruments were deployed, including a commercial instrument to measure water (G2301, Picarro Inc., Santa Clara, CA, USA). Figure 12g shows a correlation plot between the ACES Ch 455 H_2O retrievals and the Picarro data. 1 min data is shown, due to the lower time resolution of the Picarro instrument. The agree-

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ment between the two instruments is better than 5 % (slope = 0.957 ± 0.008 , $r^2 = 0.927$) and demonstrates that ACES accurately retrieves H₂O, despite its weak cross section (2.5×10^{-26} cm² at the instrument resolution, see Fig. 2).

5 Conclusions

5 The ACES instrument is engineered to provide robust, highly sensitive, and fast time-response measurements under aircraft sampling conditions. The design includes a temperature-controlled, custom-designed optical system with electronic LED on-off modulation and low-noise CCD cooled to -50°C . The instrument is also appropriate for field campaigns on other platforms, as well as for laboratory studies of trace gases and aerosols.

10 At 438–468 nm, the demonstrated precision (2σ) for field measurements of CHO-CHO and NO₂ is 34 and 80 pptv in 5 s, respectively. The calculated accuracies for these measurements are 5.8 and 5.0 %, which is consistent with standard additions and comparison to an independent NO₂ measurement.

15 At 361–389 nm, the demonstrated precision for field measurements of HONO and NO₂ is 760 and 250 pptv in 10 s, respectively. The calculated accuracies for these measurements are 9.0 and 4.6 %, which is consistent with standard additions and comparison to an independent NO₂ measurement. Due to its lower precision, the ACES HONO measurement in its current configuration is better suited to ground based sampling (e.g., 300 pptv in 1 min, 100 pptv in 10 min, 40 pptv in 1 h) and laboratory kinetic studies. In the future, higher sensitivity could be achieved through higher quality cavity mirrors or different UV light sources that better sample its strong absorption bands.

20 In addition to the target analytes, CH₃COCHO can be retrieved from fire plumes, and other high concentration environments. ACES H₂O measurements show good agreement with independent data acquired at the CARE Beijing-NCP 2014 site.

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Table 1. ACES instrumental details and measurement accuracy .

	Ch 368 (Channel 1)	Ch 455 (Channel 2)
Species	NO ₂ , HONO, O ₄	NO ₂ , CHOCHO, CH ₃ COCHO, H ₂ O, O ₄
LED power	Nichia, NCSU033B 0.45 W	LedEngin, LZ1-00DB05 1 W
Fitted wavelength range	361–389 nm	438–468 nm
Mirror reflectivity	Advanced Thin Films 99.984 % at 368 nm	Advanced Thin Films 99.9973 % at 455 nm
Cell length	48 cm	48 cm
Effective path length	3.0 km at 368 nm	17.8 km at 455 nm
Bandpass filter	Omega Optical Inc, XB05 360 ± 50 nm	Thorlabs, FB450-40 450 ± 20 nm
FWHM*	0.8 nm at 368 nm with 200 μm slit	1.02 nm at 455 nm with 200 μm slit
Accuracy	NO ₂ ±4.6 % HONO ±9.0 % O ₄ ±10.2 %	NO ₂ ±5.0 % CHOCHO ±5.8 % CH ₃ COCHO ±15.2 % H ₂ O ±4.6 % O ₄ ±10.2 %

* FWHM is determined from the Hg/Ar lamp measurement.

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Table 2. Glyoxal losses measured for different tubing materials.

Material	Surface Description	Glyoxal loss (%cm ⁻² s ⁻¹)
Aluminum – Alloy 3003	Polished	0.023 (±0.005)
Aluminum – Alloy 5052	Polished	0.072 (±0.004)
Aluminum – Alloy 6061	Polished	0.081 (±0.004)
Aluminum – Alloy 6061	Unpolished	0.141 (±0.004)
Glass		0.211 (±0.006)
Polyoxymethylene (Delrin)		0.025 (±0.004)
Stainless steel – Alloy 316		0.018 (±0.005)
Stainless steel – Alloy 316	Seamless	0.014 (±0.006)
Teflon PFA		0.0001 (±0.005)
Titanium		0.043 (±0.005)

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Table 3. Detection limit (2σ) estimation from two different methods: (1) standard deviation of gas concentrations retrieved in zero air and (2) fit errors from DOASIS. For the laboratory data, 1 h of zero air measurements in the lab were analyzed. For the field data, multiple cycles of zero air measurements acquired during CARE Beijing-NCP 2014 were analyzed.

Trace absorbers		Zero air injection (pptv)		Fit errors (pptv)	
		Lab	Field	Lab	Field
Ch 455 5 s average	NO ₂	62	80	48	36–50
	CHOCHO	30	34	30	22–32
Ch 368 10 s average	NO ₂	646	758	686	504–704
	HONO	202	254	222	250–386

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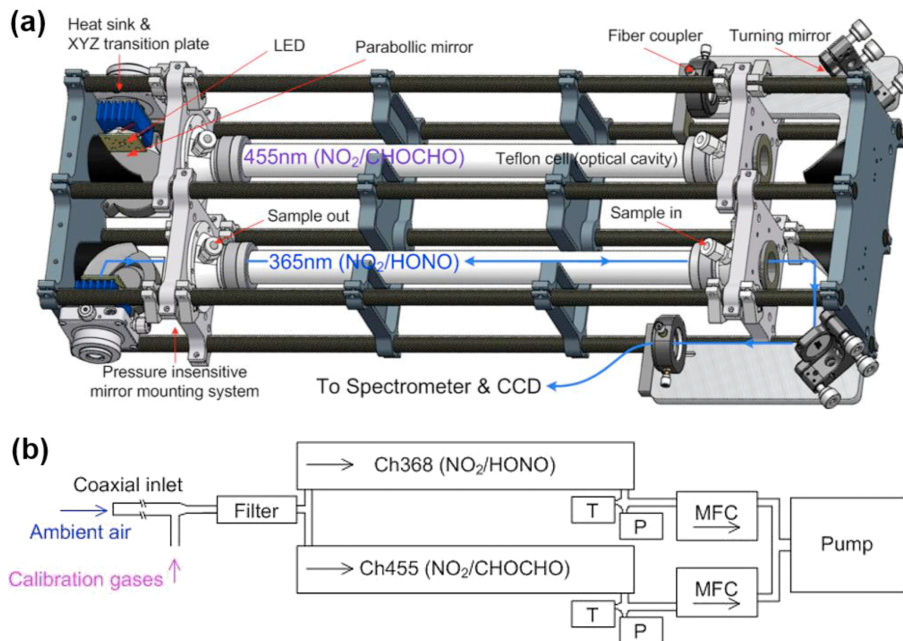


Figure 1. (a) ACES optical system, showing the LEDs, collimating optics, and high-finesse cavities defined by the high reflectivity mirrors. The system is mounted on six carbon rods to achieve an optical alignment that is insensitive to vibration and pressure changes. (b) ACES flow system, showing the inlet, aerosol filter, two cells, temperature (T) and pressure (P) sensors, mass flow controllers (MFC), and pump.

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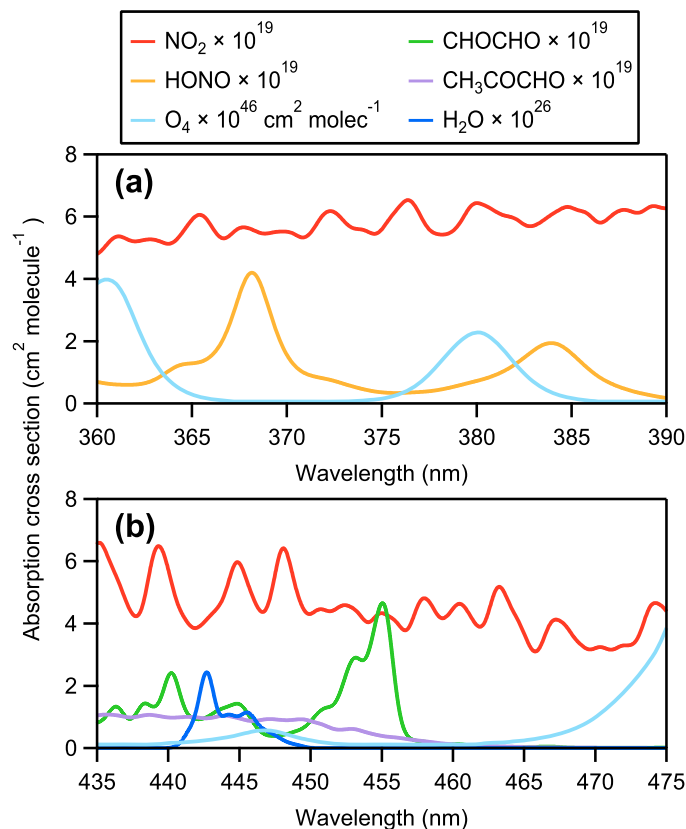


Figure 2. Convolved literature absorption cross sections shown for the ACES wavelength range for **(a)** Ch 368: NO₂, HONO, CHOCHO, O₄; **(b)** Ch 455: NO₂, CHOCHO, CH₃COCHO, H₂O, and O₄.

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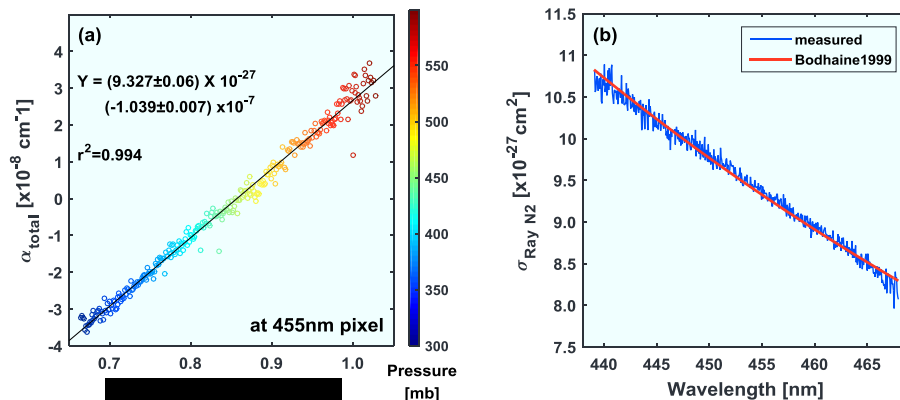


Figure 3. (a) Measured extinction at 455 nm vs. number density for a N_2 -filled cavity with varying pressure. The slope represents a Rayleigh scattering cross section ($\sigma_{\text{Rayleigh}, \text{N}_2}$) at 455 nm of $(9.33 \pm 0.06) \times 10^{-27} \text{ cm}^2$. (b) Measured wavelength-dependent $\sigma_{\text{Rayleigh}, \text{N}_2}$ at 438–468 nm compared to Rayleigh scattering cross sections reported by Bodhaine et al. (1999).

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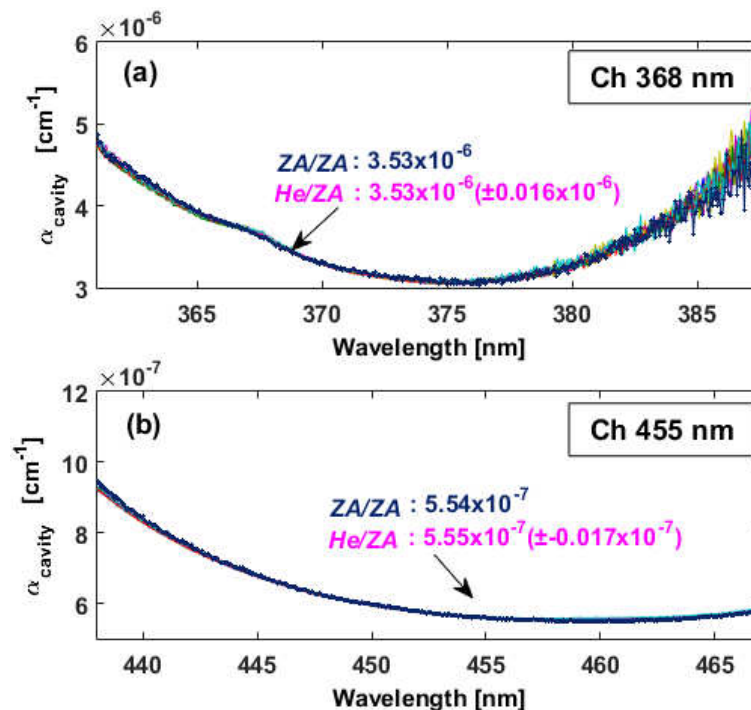


Figure 4. Cavity loss, $\alpha_{\text{cavity}}(\lambda) = (1 - R(\lambda))/d$, determined using two methods: (1) He and dry zero air (He/ZA; magenta) and (2) ZA at two different pressures (ZA/ZA; dark blue). The data are from one SENEX research flight on 5 July 2013. The mean ($\pm 1\sigma$) at 368 and 455 nm pixels for 13 repetitions of He/ZA during level flight legs are $(3.534 \pm 0.016) \times 10^{-6}$ and $(5.551 \pm 0.017) \times 10^{-7} \text{ cm}^{-1}$, respectively, shown as magenta text with varying colored lines. The estimation at those wavelengths from ZA/ZA (blue + marker and text) using 390 and 812 mb pressure (3.53×10^{-6} and 5.54×10^{-7}) are in good agreement with He/ZA.

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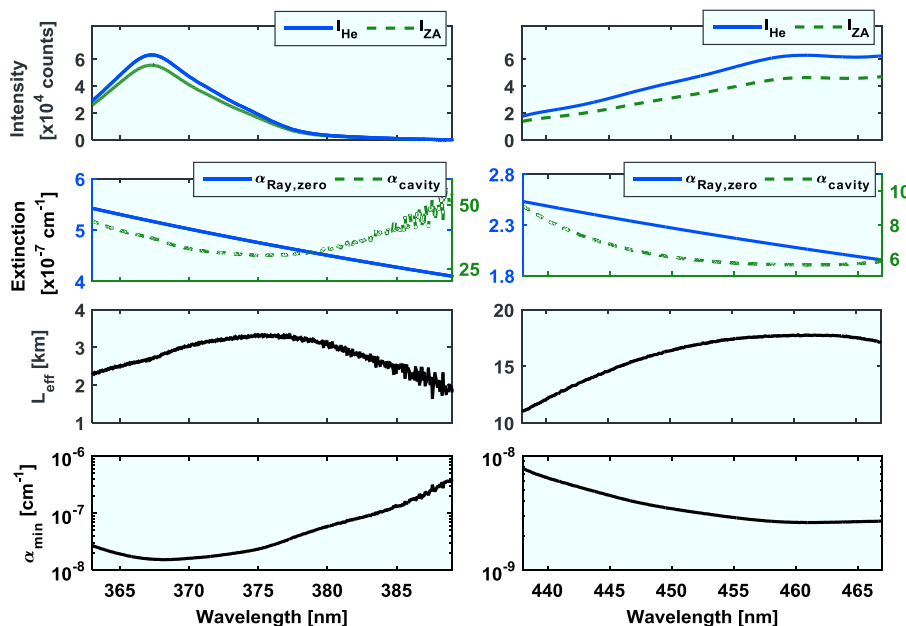


Figure 5. Results from one He/ZA measurement on 5 July 2013 with 0.5 s exposure time for Ch 368 (left panels) and Ch 455 (right panels). Figures show light intensity for He (I_{He}) and zero air (I_{ZA}) measurements for raw counts at the CCD detector (incident photon counts are $16\times$ greater due to detector gain); calculated extinction due to the Rayleigh scattering of zero air ($\alpha_{\text{Rayleigh,ZA}}(\lambda)\alpha_{\text{Ray,zero}}$); light attenuation due to cavity loss ($\alpha_{\text{cavity}}(\lambda) = (1 - R(\lambda))/d$); effective path length (L_{eff}) at each pressure (918.5 and 918.7 mb for Ch 368 and Ch 455, respectively); and calculated detection limit (α_{min}). The calculated α_{min} represent the theoretical 1σ shot noise limit for a 0.5 s exposure and do not include the factor of 16 gain setting. Theoretical α_{min} for a 5 s integration are a factor of 12.6 ($16^{1/2} \times 10^{1/2}$) lower than shown (i.e., $1.2 \times 10^{-9} \text{ cm}^{-1}$ at 368 nm in the left panel and 2.4×10^{-10} at 460 nm in the right panel).

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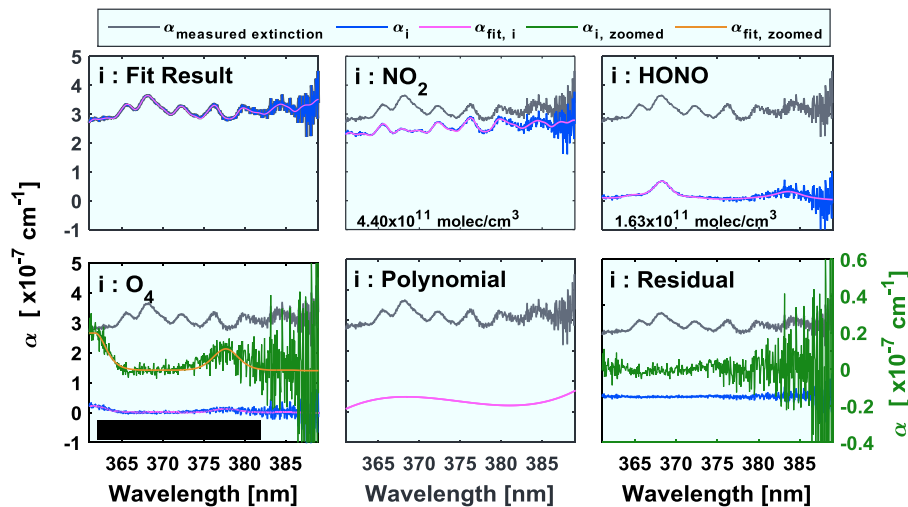


Figure 6. Example spectral fitting for one Ch 368 spectrum (10 s average) measured over an agricultural burning plume on the 3 July 2013 nighttime flight during SENEX 2013. Retrievals of NO₂, HONO, and O₄ are shown, as well as the 3rd order polynomial, total fit, and residual.

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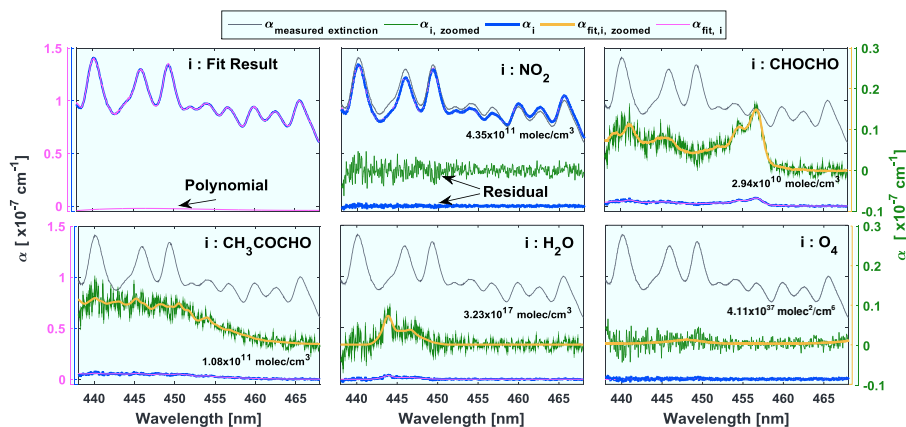


Figure 7. An example of spectral fitting for one Ch 455 (10 s average) measured over an agricultural burning plume on the 3 July 2013 nighttime flight during SENEX 2013. Retrievals of NO_2 , CHOCHO , CH_3COCHO , H_2O , and O_4 are shown, as well as the 3rd order polynomial, total fit, and residual. Blue and pink traces are plotted against the left axis, while the expanded green and yellow traces are plotted against the right axis.

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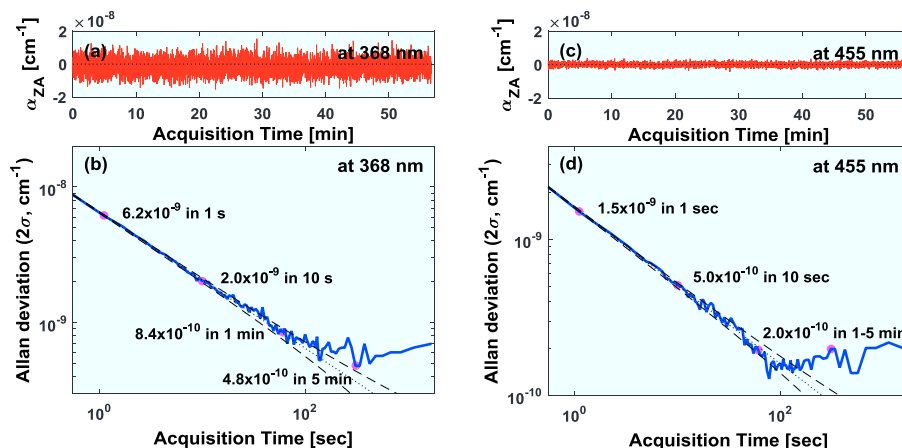


Figure 8. Allan deviation plots for single data pixels corresponding to 368 nm in Ch 368 (left) and 455 nm in Ch 455 (right). Extinction data were acquired for 1 h in zero air at constant pressure. The dotted line indicates the relationship expected for statistically random noise. Both channels have a 2 σ precision better than $5 \times 10^{-9} \text{ cm}^{-1}$ for a 10 s average and are near the theoretical limits shown in Fig. 5.

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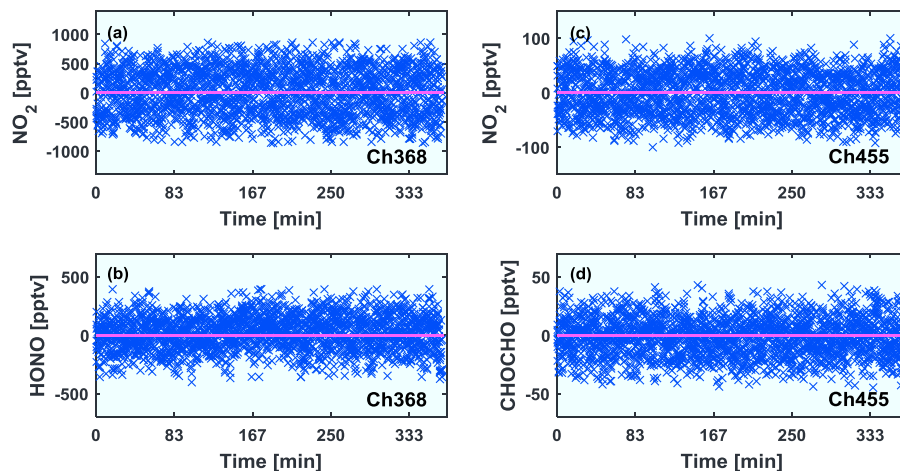


Figure 9. Spectral retrievals for Ch 368 (NO_2 and HONO; left panels) and Ch 455 (NO_2 and CHOCHO; right panels) for zero air measurements acquired during 8–14 June 2014 for CARE Beijing-NCP 2014. Each point represents a 10 s average.

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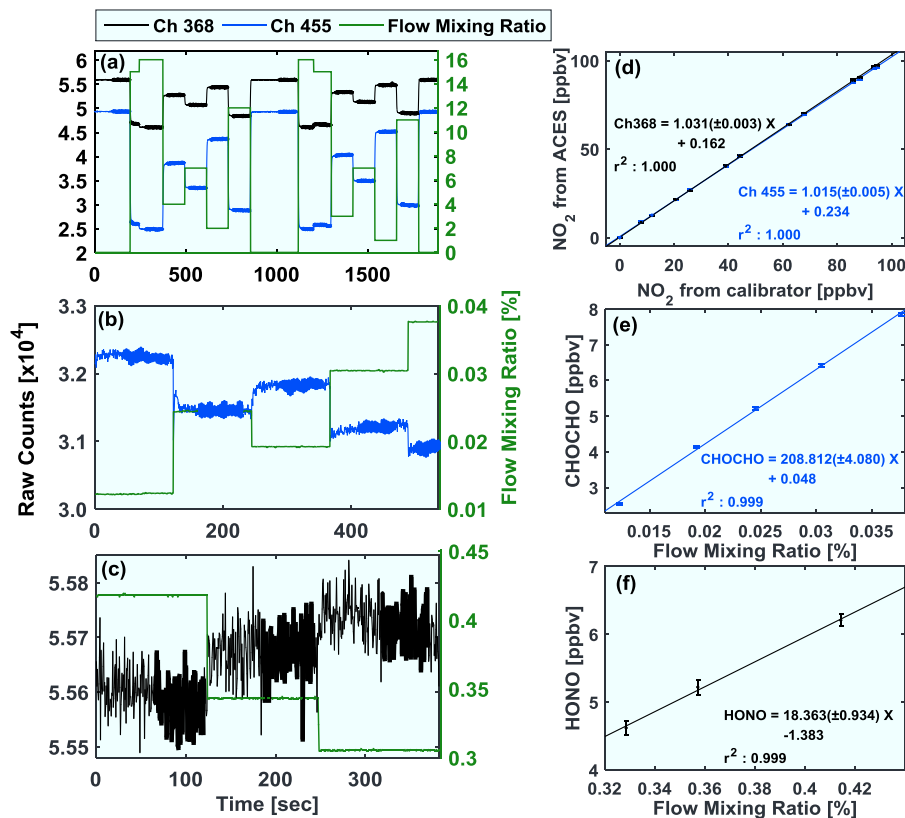


Figure 10. Standard additions of (a) NO_2 , (b) CHOCHO , and (c) HONO . Left axis shows raw data counts for a single pixel at 368 nm (Ch 368) and 455 nm (Ch 455). Right axis shows flow mixing ratio of target gas, in arbitrary units. Correlation plots for (d) NO_2 , (e) CHOCHO , and (f) HONO showing the average and standard deviation for the bold points in panels (a–c). All of the comparisons are highly linear, with $r^2 = 0.999$ or 1.000 .

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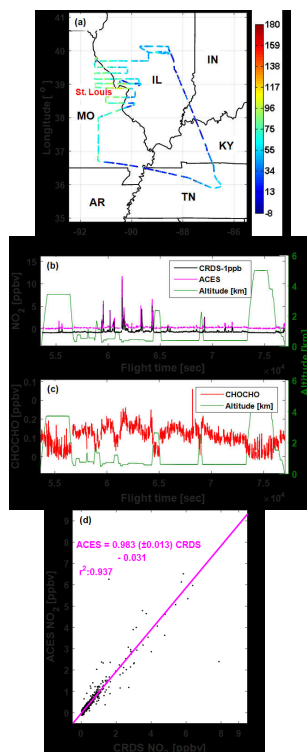


Figure 11. Data from a SENEX 2013 research flight over the central US that included areas with high isoprene and anthropogenic emissions (Ozark Mountains, MO; St. Louis, MO; Decatur, IL). **(a)** Flight map showing measured CHOCHO concentrations according to color scale. **(b, c)** Time series of NO₂ and CHOCHO, shown with aircraft altitude. NO₂ measured by CRDS is offset by 1 ppbv for clarity. **(d)** Scatter plot of ACES and CRDS NO₂ data, averaged to 10 s. The slope is 0.983 ± 0.013 with $r^2 = 0.937$.

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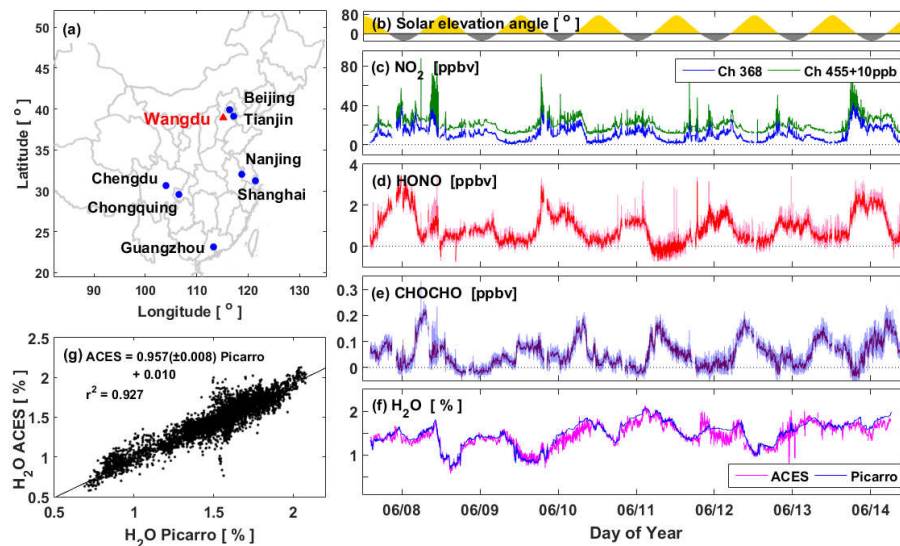


Figure 12. Data from the CARE Beijing-NCP 2014 ground site. **(a)** Map of region, showing ground site relative to major cities in China. Time series of **(b)** solar elevation angle, **(c)** NO₂, **(d)** HONO, **(e)** CHOCHO, and **(f)** H₂O. NO₂ measured by Ch 455 is offset by 10 ppbv for clarity. **(g)** Correlation plot for H₂O measured by ACES and a Picarro instrument. The slope is 0.957 ± 0.008 with $r^2 = 0.927$.

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